

AD-A125 786

COMPARISON OF DATA FROM THE LOW ENERGY ELECTROSTATIC
ANALYZERS ON SATELLITE P78-1(U) AIR FORCE GEOPHYSICS
LAB HANSCOM AFB MA R P VANCOU 19 AUG 82

1/1

UNCLASSIFIED

AFGL-TR-82-8236

F/G 20/3

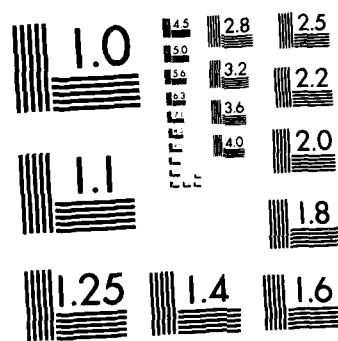
NL

END

FILED

19

82



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS 1963-A

(12)

125706

AFGL-TR-82-0236
ENVIRONMENTAL RESEARCH PAPERS, NO. 790



Comparison of Data From the Low Energy Electrostatic Analyzers on Satellite P78-1

ROGER P. VANCOUR

19 August 1982

Approved for public release; distribution unlimited.

SPACE PHYSICS DIVISION
AIR FORCE GEOPHYSICS LABORATORY
HANSCOM AFB, MASSACHUSETTS 01731

PROJECT 7661

AIR FORCE SYSTEMS COMMAND, USAF

83 03 15 015



This report has been reviewed by the ESD Public Affairs Office (PA)
and is releasable to the National Technical Information Service (NTIS).

This technical report has been reviewed and
is approved for publication.

Alva T. Stair, Jr.

DR. ALVA T. STAIR, Jr.
Chief Scientist

Qualified requestors may obtain additional copies from the
Defense Technical Information Center. All others should apply
to the National Technical Information Service.

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER AFGL-TR-82-0236	2. GOVT ACCESSION NO. AD-A125706	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) COMPARISON OF DATA FROM THE LOW ENERGY ELECTROSTATIC ANALYZERS ON SATELLITE P78-1	5. TYPE OF REPORT & PERIOD COVERED Scientific. Interim.	
7. AUTHOR(s) Roger P. Vancouver	6. PERFORMING ORG. REPORT NUMBER ERP No. 790	
9. PERFORMING ORGANIZATION NAME AND ADDRESS Air Force Geophysics Laboratory (PHK) Hanscom AFB Massachusetts 01731	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS 62101F 76611003	
11. CONTROLLING OFFICE NAME AND ADDRESS Air Force Geophysics Laboratory (PHK) Hanscom AFB Massachusetts 01731	12. REPORT DATE 19 August 1982	
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)	13. NUMBER OF PAGES 11	
	15. SECURITY CLASS. (of this report) Unclassified	
	15a. DECLASSIFICATION/DOWNGRADING SCHEDULE	
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Electrostatic analyzers Energy levels Electrons		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Air Force satellite P78-1 has aboard, on the wheel section, two detectors each consisting of a low energy and a high energy electron electrostatic analyzer. The low energy analyzer collects electrons in the energy range of 50 eV to 1 keV and the high energy analyzer collects electrons in the energy range of 1 keV to 20 keV. The two detectors were placed 90° apart on the wheel section of the satellite. The leading detector in the wheel's rotation is denoted as Detector 1.		

DD FORM 1473

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

20. (Contd)

Counts of low energy electrons made by the low energy analyzers on board the satellite do not agree, but can be corrected so that all data are usable. Multiplying the readings of the low energy analyzer of Detector 1 by a factor of 1.64 brings them into agreement with counts from the other low energy analyzer and the two high energy analyzers. Three different comparisons of data from the four analyzers were used to derive the correction factor.

Unclassified

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

Contents

1. DESCRIPTION OF THE PROBLEM	5
2. EXPERIMENT AND INSTRUMENTS	6
2.1 Satellite Orbit	6
2.2 Instruments	6
3. COMPARISON OF RESULTS	6
3.1 J_{tot} Values, Channels 1-8, Both Detectors	7
3.2 Comparison of Each Energy Channel (1-16) of the Two Detectors	8
3.3 $\frac{dJ}{dE}$ Values for Channels 8 and 9, Both Detectors	10
4. CONCLUDING REMARKS	10

Illustrations

1. Electron Differential Energy Spectra of All Channels From Both Detectors

Accession For	
NTIS GRA&I <input checked="" type="checkbox"/>	
DTIC TAB <input type="checkbox"/>	
Unannounced <input type="checkbox"/>	
Justification	
By	
Distribution /	
Available in Other	
Dist	Author
Dist	Special
A	



Tables

1. Average Values of $\frac{J_{tot} (2)}{J_{tot} (1)}$ for Several Orbits	8
2. Ratio of Counts for Each Energy Channel, Both Detectors	9
3. Comparison of Channels 8 and 9, Both Detectors	10

Comparison of Data From the Low Energy Electrostatic Analyzers on Satellite P78-1

I. DESCRIPTION OF THE PROBLEM

As previously reported,¹ the low energy analyzer of Detector 1 on satellite P78-1 consistently returns significantly lower counts than the low energy analyzer of Detector 2. This has been observed in the reduced data from many orbits. The problem does not exist in the high energy analyzers.

The four analyzers had been tested with four tritium radioactive sources prior to launch. Two sources were mounted on each of two holders in such a way that the sources fit exactly over the slits of each analyzer of the two detectors. Tests were run prior to satellite integration to compare the counts observed by each analyzer. The same type of tests were run on two other detectors which were also calibrated² at Rice University, Houston, Texas. In all cases, the response of the analyzer for the same energy range agreed within 20%. The problem can be due to an inaccuracy in one or both channeltrons of the low energy analyzer of Detector 1 or in the electron geometry of that analyzer. In either case, the geometric factors

¹ Report dated 10 August 1962.

² Vondrak, R. M. "1964 High Latitude Electron Detectors on Satellite P78-1: Preliminary Data," AFGL-TR-64-0131, AD A104518.

² Hines, D. J., Jr., Semenov, M. S., and Huber, A. (1970) The Precipitating Electron Detectors (SSA-3) for the Block 5D-Flights 2-5 DMSP Satellite: Preliminary Preparation, AFGL-TR-70-0210, AD A082133.

can be corrected by comparing the four analyzers and the data is useful. The problem is resolved by making such comparisons.

2. EXPERIMENT AND INSTRUMENTS

2.1 Satellite Orbit

Satellite P78-1, launched 25 February 1979, carries high latitude electron detectors. The satellite is in a circular, polar, sun-synchronous orbit (noon-midnight meridian plane) at an altitude of 600 ± 30 km, with an inclination of 97.73° . The satellite consists of a wheel section, carrying the instruments, and a sail section, carrying the photocells that power the satellite. The wheel section spins at 11 ± 1 rpm, so that the spin period is 5.45 ± 0.150 sec. The spin axis of the wheel section is perpendicular to the orbital plane of the satellite.

2.2 Instruments

Two electron detectors are mounted in a single container so their look directions are 90° apart. Both look outward from the rim of the wheel section. The spin of the satellite scans the detectors through all pitch angles in each spin period. The two detectors are identical to those flown on the DMSP satellites. The calibrations used for the detectors are found in Hardy et al.² The calibrations used are an average of the values for Detectors 4 and 5 in that work. Each detector consists of two curved plate electrostatic analyzers with channeltrons. One analyzer of each detector collects electrons in the energy range of 50 eV to ~ 1 keV and the other analyzer of each detector collects electrons in the energy range of 1-20 keV. Each analyzer has eight energy channels. All four analyzers are stepped through the eight channels simultaneously every 256 msec. Thus, each set of analyzers produces four 16-point spectra in just over 1 sec or about 22 spectra in each vehicle rotation. The lead detector in the vehicle rotation is called Detector 1 and the detector following by 90° is called Detector 2.

3. COMPARISON OF RESULTS

Since the two detectors are 90° apart, a comparison can best be made when both of them are near $+45^\circ$ pitch angle in the northern hemisphere or both are near -135° pitch angle in the southern hemisphere.

Data are used from four different satellite orbits, namely, 220, 438, 530, and 711, and only in the auroral regions when the counting is sufficiently high. Data are taken in four 45° pitch angle bins:

- a. When the stated values of the pitch angles of the two detectors, pa_1 and pa_2 , differ by 2° or less,
- b. When the difference is greater than 2° but less than or equal to 10° , and,
- c. When the difference is greater than 10° but less than or equal to 20° .

Three comparisons are made. The integrated number flux, J_{tot} , is used to compare the two low energy analyzers (Channels 1-8). Whenever J_{tot} for the eight channels was less than $2 \times 10^7 \text{ e cm}^{-2} \text{ sec}^{-1} \text{ sr}^{-1}$, the counts were very low, gave poor statistics, and so were not used. The second comparison is of each channel (1-8) of one detector with the corresponding channel of the other detector. The third comparison uses the differential flux values, dJ/dE , of both detectors for Channels 8 and 9. Channel 8 is centered at 0.974 keV and Channel 9 at 1.000 keV. Since they are very nearly at the same energy level, the differential fluxes can be compared.

3.1 J_{tot} Values, Channels 1-8, Both Detectors

Data from all four orbits are used for each pitch angle bin. Orbits 530 and 959 occurred during fairly active K_p periods while orbits 220 and 438 occurred during moderate K_p periods.

In this section J_{tot} is calculated for each time period for both detectors by

$$J_{tot} = \sum_{i=1}^8 \frac{dJ}{dE_i} \Delta E_i. \quad (1)$$

As can be seen in Eq. (1), only Channels 1-8 are used in the determination of J_{tot} since a comparison is being made of the low energy analyzers only. The ratio of Detector 2 to Detector 1 values is written as $\frac{J_{tot}^{(2)}}{J_{tot}^{(1)}}$ and is determined. Values are found for all three pitch angle bins and the average of all values in each bin and in each orbit are calculated. Table 1 lists the results showing the number of samples, n ; the average value of the ratio $\frac{J_{tot}^{(2)}}{J_{tot}^{(1)}}$, \bar{x} ; and the standard deviation of the ratio, s .

The results of all pitch angle bins and all four orbits are in good agreement and give an overall average value for all cases of $\frac{J_{tot}^{(2)}}{J_{tot}^{(1)}} = 1.640$ for 314 samples.

Table 1. Average Values of $\frac{J_{tot}(2)}{J_{tot}(1)}$ for Several Orbits
 $(J_{tot}$ Values are for Channels 1-8 Only)

Case A, $ pa_1 - pa_2 \leq 2^\circ$					
Orbit	220	438	530	959	Total
n	12	4	17	2	35
\bar{x}	1.614	1.651	1.578	1.425	1.590
s	0.182	0.079	0.272	0.093	0.221
Case B, $2^\circ < pa_1 - pa_2 \leq 10^\circ$					
n	35	14	68	19	136
\bar{x}	1.650	1.622	1.650	1.682	1.651
s	0.179	0.165	0.209	0.224	0.198
Case C, $10^\circ < pa_1 - pa_2 \leq 20^\circ$					
n	43	13	69	18	143
\bar{x}	1.606	1.648	1.638	1.734	1.642
s	0.194	0.172	0.361	0.274	0.294
Case A + B + C					
n	90	31	154	39	314
\bar{x}	1.624	1.637	1.637	1.693	1.640
s	0.186	0.156	0.292	0.249	0.248

3.2 Comparison of Each Energy Channel (1-16) of the Two Detectors

As a second technique, we compared the counts for each energy channel of the two detectors. Table 2 shows the ratio of the counts from Detector 2 to the counts of Detector 1 for each of the 16 energy channels. Eighty-eight samples were used from Orbit 220 for Channels 1-8 and 40 samples were used for Channels 9-16. It can be seen that in the lower energy analyzers, the lowest average ratio is in Channel 7 (1.579) and the highest value is in Channel 4 (1.655). The average for all values of Channels 1-8 is 1.619. The ratios in Channels 9-16 are from 0.976 to 1.086 and the average ratio for these channels is 1.022. Thus, the two mid-energy analyzers are in excellent agreement.

Figure 4 shows a sample of the differential spectra for the 16 channels of the two detectors. It can be seen that the low-energy portion of each of the two curves are not in agreement. By correcting the $\frac{d^2p}{dp d\theta}$ values of Channel 1 of Detector 1 by multiplying each value by 1.619 gives the eight counts in Channel 1

third spectrum. This gives good agreement with the lower energy portion of the spectrum from Detector 2.

Table 2. Ratio of Counts for Each Energy Channel, Both Detectors

	Cts (Det 2) Cts (Det 1)		Cts (Det 2) Cts (Det 1)
Ch 1	1644 ± 0.165	Ch 9	1.017 ± 0.189
2	1.652 ± 0.234	10	0.989 ± 0.120
3	1.609 ± 0.267	11	1.000 ± 0.137
4	1.655 ± 0.160	12	1.043 ± 0.189
5	1.594 ± 0.159	13	1.006 ± 0.148
6	1.592 ± 0.199	14	1.020 ± 0.176
7	1.579 ± 0.242	15	1.011 ± 0.180
8	1.626 ± 0.157	16	1.086 ± 0.223

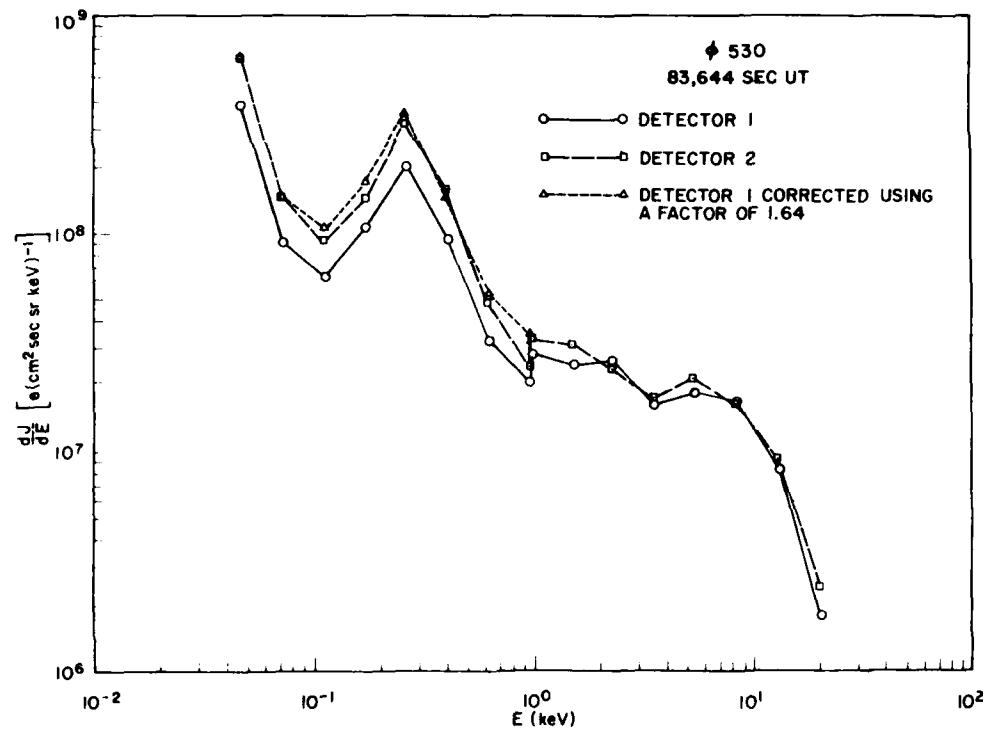


Figure 1. Electron Differential Energy Spectra of All Channels From Both Detectors

3.3 $\frac{dJ}{dE}$ Values for Channels 8 and 9, Both Detectors

Orbits 220 and 530 are used to obtain counts for Channels 8 and 9 from both detectors. Thirty samples are from orbit 220 and 40 samples from orbit 530.

Since Channel 8 is centered at 0.974 keV and Channel 9 at 1,000 keV, they should count nearly alike. In addition, Channel 8 on each detector should be nearly equal, as should Channel 9 on each detector. The question arises as to whether the low energy analyzer of Detector 1 is counting low or that of Detector 2 is counting high. This question can be resolved by these comparisons.

It can be seen in Table 3 that the ratio $\left(\frac{8_2}{8_1}\right)$ of the differential flux values, dJ/dE , of both channels is 1.639 and the ratio $\left(\frac{9_1}{8_1}\right)$ of the dJ/dE values is 1.611. Both of these values agree very well with the overall average value of 1.640 from Table 1. In addition, the ratios, $\left(\frac{9_2}{8_1}\right)$ and $\left(\frac{9_2}{8_2}\right)$ of the differential fluxes are close to unity. Therefore, Channels 8 and 9 on Detector 2 agree well with each other and with Channel 9 on Detector 1. However, each of them is higher than Channel 8 of Detector 1 by a factor of about 1.600. This indicates that the low energy analyzer Channels 1 through 3 of Detector 1 give counts that are low by this factor.

Table 3. Comparison of Channels 8 and 9, Both Detectors

$\frac{dJ}{dE}$ Ratio	Rev 220	Rev 530	Overall
$8_2/8_1$	1.615	1.650	1.639
$9_2/8_1$	1.011	1.079	1.057
$9_1/8_1$	1.712	1.567	1.611
$9_2/8_2$	1.072	1.025	1.039

4. CONCLUDING REMARKS

It was observed in the early data from the electrostatic analyzers of satellite P73-1 that the dJ/dE values for Detector 1 were lower than those of Detector 2, except when Detector 1 was in a position relative to Detector 2 to receive more solar electron counts. It was observed that the counts from the front channels of the Detector 1

low energy analyzer were substantially lower than those from the eight channels of the Detector 2 low energy analyzer. The high energy analyzers were in good agreement in all channels (9-16).

In discussions with the designers and builders of the instruments,^{3, 4} the conclusion was reached that either one or both channeltrons in the low energy analyzer of Detector 1 was inefficient and gave a lower count than it should, or the problem was in the electronic circuitry of that analyzer which made the counting inefficient in all eight channels. In either case, the geometric factor of each of the eight energy channels was affected, causing a lower count than normal by a certain factor. In this report, the correction factor for the low energy analyzer data of Detector 1 is determined, to make all data useful.

Several orbits of data are analyzed and each analyzer and energy channel are compared when the two detectors are at or near the same pitch angle value, at which time they should read alike.

Three comparisons are made and all agree that the counts for all eight channels of the low energy analyzer of Detector 1 should be corrected by multiplying the counts of that analyzer by 1.6. If this is done, then the calculations of the differential flux dJ/dE , the integrated number flux J_{tot} , and the integrated energy flux J_{Etot} , will be performed in the same manner that they are for the other analyzers. The data are usable but must be corrected as just noted.

The data used in these calculations came from orbits that were early in the flight of P78-1. Some orbits, from several months later, will be analyzed to see if this correction factor for the low energy analyzer of Detector 1 remains the same or has changed. If the correction factor has changed, it will have to be updated at various intervals during the flight of P78-1.

3. Pantazis, J., Huber, A., Hagan, M. P. (1977) Design of Electrostatic Analyzer, AFGL-TR-77-0120, AD 042564.
4. Huber, A., Pantazis, J., Besse, A. L., and Rothwell, P. L. (1977) Calibration of the SSJ/3 Sensor on the DMSP Satellites, AFGL-TR-77-0202, AD A045997.